

Numerical and experimental investigation of sound propagation in porous media for aeronautics

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Acoustic liner: micro perforated panel.

Context of the internship

Single perforation: (r, L_{cav}, ϕ, d) .





- Understanding the sound damping system of a turbofan.
- Acoustic response at high sound pressure level under normal incidence.



• Studying the micro-structure.

Model

5 parameters porous media description: (σ , ϕ , α_{∞} , Λ , Λ'). Acoustic surface impedance:

$$Z(\omega) = \frac{p(\omega)}{\mathbf{v}(\omega) \cdot \mathbf{n}_{\mathsf{S}}} = R(\omega) + j\chi(\omega).$$







Numerical & experimental results

Harmonic excitation

• Computations: 110dB to 170dB / 1600, 2500 & 4000 Hz.

Radius <i>r</i>	Cavity length L _{cav}	Porosity ϕ	Width <i>d</i>
0.15 mm	20 mm	5%	0.8 mm

Vortex shedding absorption mechanism.

Velocity magnitude (m/s) 0.00 8.33 16.7 25.0 33.3 41.7 50.0 58.3 66.7 75.0 Velocity magnitude (m/s) 16.7 25.0 33.3 41.7 50.0 58.3 66.7 75.0

Broadband excitation

Excitation by a linear chirp signal.





Fig. 2 – Time evolution of velocity and vorticity fields at 160 dB and 4000 Hz. Arrows: flow direction. Contours: Q-criterion. $\Delta t = 1/20$ T.

Fig. 3 – Acoustic properties of the perforated panel for different sound pressure levels. AlphaCell model and ProLB comparison.

 \blacktriangleright Measurements in an impedance tube \Rightarrow good agreement.

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