Non-reflecting boundary conditions and domain decomposition methods for industrial flow acoustics

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1. Industrial context

Physical models for sound propagation Reaching the memory limit Objective of the thesis

2. Domain decomposition framework

Method overview Flow acoustics formulation Updated objective

- 3. ABCs for heterogeneous and convected problems Microlocal analysis Application to the Linearized Potential Equation Numerical examples
- 4. Application to non-overlapping domain decomposition Heterogeneous waveguide problems Convected problem in freefield
- 5. The 3D turbofan intake radiation problem Numerical results for the turbofan problem Solver weak scalability



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Industrial context

Long term perspective

Predict noise from bodies in motion for the transport industry

Computational (aero)acoustics



- 1. analyze & extract sources
- 2. understand sound propagation
- 3. find solutions (new material or design)

Industrial objective

Provide a "*ready-to-use*" sound propagation simulation tool

- suitable to modern computer architectures
- applicable to large, complex industrial problems
- \rightarrow can serve as basis for optimization routines

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Physical models for sound propagation



Hybrid model - solve mean flow and acoustic perturbations separately

- we choose the time-harmonic Linearized Potential Equation
- simple but accurate for single tones of turbofan engine intakes

Physical model - Linearized Potential Equation

Scalar equation for the acoustic velocity potential $oldsymbol{v}=
abla u$

Linearized Potential Equation (LPE)

$$\rho_0(\boldsymbol{x})\frac{\mathrm{D}_0}{\mathrm{D}t}\left(\frac{1}{c_0(\boldsymbol{x})^2}\frac{\mathrm{D}_0\boldsymbol{u}}{\mathrm{D}t}\right) - \nabla\cdot\left(\rho_0(\boldsymbol{x})\nabla\boldsymbol{u}\right) = \boldsymbol{f}, \quad \frac{\mathrm{D}_0}{\mathrm{D}t} = \mathrm{i}\boldsymbol{\omega} + \boldsymbol{v}_0(\boldsymbol{x})\cdot\nabla$$

Helmholtz-type problem with convection and heterogeneities

Mathematical difficulties

- oscillatory, non-local solution
- complex valued, strongly indefinite with ω
- unbounded domain
- convection effects

does not converge with classical iterative methods [*Ernst, Gander 2012*]

 \rightarrow use a direct solver

Point source in a uniform flow



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Reaching the high frequency limit

Industrial example : single tone turbofan intake radiation **Current solver :** high-order *p*-FEM with direct solver (MUMPS)

 $\omega_{\rm bpf} \leftrightarrow \approx 25$ wavelengths -0.05 0.05

 $\omega_{\rm bpf}$, $N_{\rm dofs} = 10$ M, nnz = 730 M Direct solver \rightarrow 740 Gb of RAM

 \downarrow increase ω ?

 $2 \times \omega_{bpf}$, $N_{dofs} = 73$ M, nnz = 5B Direct solver ≈ 6 Tb of RAM ...

 $\mathcal{O}(\omega^3)$ scaling in memory & time ...

can we distribute the memory cost ? \rightarrow domain decomposition

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Objective of the thesis

Industrial objective

Provide a (scalable) parallel solver to increase the upper frequency limit

Starting point of the thesis

Discretization

- high-order finite elements

 → reduce discretization error
 (interpolation & dispersion)
- adaptive order based on *a-priori* error indicator [*Bériot et al. 2016*]
 → less unknowns

Parallelization

- algebraic parallelization is hard for Helmholtz problems
- instead, "divide and conquer" at the continuous (PDE) level → domain decomposition
- lots of approaches, but common framework [*Gander, Zhang 2019*]

Selected solution - 1st objective

Extend the non-overlapping optimal Schwarz domain decomposition framework [Boubendir et al. 2012] to the Linearized Potential Equation

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Toy example: disk scattering by a plane wave



Toy example: disk scattering by a plane wave



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Toy example: disk scattering by a plane wave

 Partition Ω = U^{N_{dom}-1}_{i=0} Ω_i into subdomains

Iterate until convergence

1. Solve the volume subproblems *u_i* with boundary conditions



Toy example: disk scattering by a plane wave

 Partition Ω = U^{N_{dom}-1}_{i=0} Ω_i into subdomains

Iterate until convergence

- 1. Solve the volume subproblems *u_i* with boundary conditions
- 2. update the interfaces unknowns $\boldsymbol{g} = (g_{ij}, g_{ji})$ through transmission conditions $(\mathcal{S}_i, \mathcal{S}_j) \Leftrightarrow$ solve surface problem



Toy example: disk scattering by a plane wave

• Partition $\Omega = \bigcup_{i=0}^{N_{dom}-1} \Omega_i$ into subdomains

Iterate until convergence

- 1. Solve the volume subproblems *u_i* with boundary conditions
- 2. update the interfaces unknowns $\mathbf{g} = (g_{ij}, g_{ji})$ through transmission conditions $(\mathcal{S}_i, \mathcal{S}_j) \Leftrightarrow$ solve surface problem
 - convergence ? [Desprès 1991]
 - How to choose the operators (S_i, S_j) ? \rightarrow numerous works ...



Optimal convergence with the **Dirichlet-to-Neumann** operator

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Flow acoustics formulation

In each subdomain Ω_i , solve the boundary value problems

Non-overlapping optimal Schwarz formulation

$$\begin{array}{l} \rho_0 \frac{D_0}{Dt} \left(\frac{1}{c_0^2} \frac{D_0 u_i}{Dt} \right) - \nabla \cdot (\rho_0 \nabla u_i) = 0 \text{ in } \Omega_i, \text{ (LPE)} \\ \rho_0 \left(1 - M_n^2 \right) \left(\partial_{n_i} u_i + i \widetilde{\Lambda}^+ u_i \right) = 0, \text{ on } \Gamma_i^\infty \text{ (radiation condition)} \\ \rho_0 \left(1 - M_n^2 \right) \left(\partial_{n_i} u_i + i \mathcal{S}_i u_i \right) = g_{ij}, \text{ on } \Sigma_{ij}, \text{ (interface condition)} \end{array}$$

Introduce the interface coupling on Σ_{ij}

$$g_{ij} = \rho_0 \left(1 - M_n^2 \right) \left(-\partial_{n_j} u_j + i S_i u_j \right) \\ = -g_{ji} + i \rho_0 \left(1 - M_n^2 \right) \left(S_i + S_j \right) u_j := \mathcal{T}_{ji} g_{ji} + b_{ji}$$

Rewrite the coupling as a linear system for $\boldsymbol{g} = (g_{ij}, g_{ji})^{\mathcal{T}}$



 \mathcal{T}_{ij} and \mathcal{T}_{ji} are the iteration operators, and can be written in terms of $\widetilde{\Lambda}^+$ PhD defense June 16, 2022 17 / 60

High-level algorithmic procedure

Surface iteration operators

$$\mathcal{T}_{ji} = rac{\mathcal{S}_i - \widetilde{\Lambda}^+}{\mathcal{S}_j + \widetilde{\Lambda}^+}, \quad \mathcal{T}_{ij} = rac{\mathcal{S}_j + \widetilde{\Lambda}^-}{\mathcal{S}_i - \widetilde{\Lambda}^-}$$

Iteration matrix eigenvalues: $\lambda_{(I-A)} = 1 \pm \sqrt{\mathcal{T}_{ji}\mathcal{T}_{ij}}$ If we choose $S_i = \tilde{\Lambda}^+$ and $S_j = -\tilde{\Lambda}^-$, we have optimal convergence

Parallel iterative algorithm for the process iDo in Ω_i at iteration (n + 1), $\forall j \in D_i$

- 1. given $g_{ij}^{(n)}$, solve $u_i^{(n+1)}$ in Ω_i ,
- 2. update the (n + 1) neighbourhood data through $g_{ji}^{(n+1)} = -g_{ij}^{(n)} + i\rho_0 \left(1 M_n^2\right) (S_i + S_j) u_i^{(n+1)}$ on Σ_{ij} ,

High-level algorithmic procedure

Surface iteration operators

$$\mathcal{T}_{ji} = rac{\mathcal{S}_i - \widetilde{\Lambda}^+}{\mathcal{S}_j + \widetilde{\Lambda}^+}, \quad \mathcal{T}_{ij} = rac{\mathcal{S}_j + \widetilde{\Lambda}^-}{\mathcal{S}_i - \widetilde{\Lambda}^-}$$

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 $(\widetilde{\Lambda}^+, -\widetilde{\Lambda}^-)$ are **non-local** DtN maps for the LPE \rightarrow design **sparse approximations** $S_i \approx \widetilde{\Lambda}^+$ and $S_j \approx -\widetilde{\Lambda}^ \Leftrightarrow$ approximate Schur complements at the algebraic level

Algebraic interpretation of domain decomposition

Global problem for two subdomains (i, j) with a common interface Σ

$$\begin{pmatrix} \mathbb{K}_{i}^{\Omega} & 0 & \mathbb{K}_{i}^{\Omega,\Sigma} \\ 0 & \mathbb{K}_{j}^{\Omega} & \mathbb{K}_{j}^{\Omega,\Sigma} \\ \mathbb{K}_{i}^{\Sigma,\Omega} & \mathbb{K}_{j}^{\Sigma,\Sigma} & \mathbb{K}_{i}^{\Sigma,\Sigma} + \mathbb{K}_{j}^{\Sigma,\Sigma} \end{pmatrix} \begin{pmatrix} u_{i}^{\Omega} \\ u_{j}^{\Omega} \\ u^{\Sigma} \end{pmatrix} = \begin{pmatrix} f_{i}^{\Omega} \\ f_{j}^{\Omega} \\ f_{i}^{\Sigma} + f_{j}^{\Sigma} \end{pmatrix}$$

Direct parallel solver \rightarrow block LU factorization per subdomain

$$\begin{pmatrix} \mathbb{I} & 0 & 0 \\ 0 & \mathbb{I} & 0 \\ \mathbb{K}_{i}^{\Sigma,\Omega}(\mathbb{K}_{i}^{\Omega})^{-1} & \mathbb{K}_{j}^{\Sigma,\Omega}(\mathbb{K}_{j}^{\Omega})^{-1} & \mathbb{I} \end{pmatrix} \begin{pmatrix} \mathbb{K}_{i}^{\Omega} & 0 & 0 \\ 0 & \mathbb{K}_{j}^{\Omega} & 0 \\ 0 & 0 & \mathbb{L} \end{pmatrix} \begin{pmatrix} \mathbb{I} & 0 & (\mathbb{K}_{i}^{\Omega})^{-1} \mathbb{K}_{i}^{\Omega,\Sigma} \\ 0 & \mathbb{I} & (\mathbb{K}_{j}^{\Omega})^{-1} \mathbb{K}_{j}^{\Omega,\Sigma} \\ 0 & 0 & \mathbb{I} \end{pmatrix}$$

The Schur complement $\mathbb{L} = \mathbb{L}_i + \mathbb{L}_j$ is dense \Leftrightarrow discrete DtN map \rightarrow advances on dense Block Low-Rank factorization [*Amestoy et al.* 2019]

The non-overlapping Schwarz approach can be seen as an iterative solver for \mathbb{L} preconditioned by interfaces conditions

 \rightarrow approximate block LU factorization at the continuous level

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Summary and updated objective

Non-overlapping domain decomposition for flow acoustics

- similar to the Helmholtz formulation, common framework
- quick convergence relies on sparse approximations of the DtN map
 ⇔ Non-reflecting boundary conditions

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Updated objective

Design non-reflecting boundary conditions for heterogeneous and convected time-harmonic problems

Two techniques were studied during the thesis

- 1. Absorbing Boundary Conditions (ABC) [Marchner et al. SIAP 2022],
- 2. Perfectly Matched Layers (PML) [Marchner et al. JCP 2021],

Summary and updated objective

Non-overlapping domain decomposition for flow acoustics

- similar to the Helmholtz formulation, common framework
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Next section

Focus on the construction of Absorbing Boundary Conditions

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Microlocal analysis - DtN operator

Goal: find local approximations to the Dirichlet-to-Neumann map

DtN operator on $\boldsymbol{\Sigma}$

$$\widetilde{\Lambda^{+}}: \begin{cases} H^{1/2}(\Sigma) \to H^{-1/2}(\Sigma) \\ u_{|\Sigma} \mapsto \partial_{\mathbf{n}} u_{|\Sigma} = -i\widetilde{\Lambda^{+}} u_{|\Sigma} \end{cases}$$

through pseudo-differential calculus

[Engquist and Majda 1977, 1979] [Antoine et al. 1999]

Example : 2D heterogeneous Helmholtz half-space problem

$$\mathcal{L} = \rho_0^{-1} \partial_x (\rho_0 \partial_x) + \rho_0^{-1} \partial_y (\rho_0 \partial_y) + \omega^2 c_0^{-2}$$

$$\approx \left(\partial_x + i \sqrt{\omega^2 c_0^{-2} + \rho_0^{-1} \partial_y (\rho_0 \partial_y)} \right) \left(\partial_x - i \sqrt{\omega^2 c_0^{-2} + \rho_0^{-1} \partial_y (\rho_0 \partial_y)} \right)$$

We cannot formally factorize \mathcal{L} when $\partial_x(\rho_0) \neq 0$ or $\partial_x(c_0) \neq 0$

- Apply the principle to the **symbol** λ^+ of Λ^+
- ightarrow work on polynomials in the co-tangent Fourier space ξ



Pseudo-differential operator

We define a differential operator of order m,

$$\mathcal{P}(\mathbf{x},\partial_{\mathbf{x}}) = \sum_{|\boldsymbol{\alpha}| \leq m} (-\imath)^{\boldsymbol{\alpha}} \boldsymbol{a}_{\boldsymbol{\alpha}}(\mathbf{x}) \partial_{\mathbf{x}}^{\boldsymbol{\alpha}}, \quad \mathbf{x} \in \mathbb{R}^{d}, \ \boldsymbol{\alpha} = (\alpha_{1},\ldots,\alpha_{d}) \in \mathbb{N}^{d},$$

through its inverse Fourier representation \rightarrow more general framework

Pseudo-differential operator of order m

$$\mathcal{P}(\mathbf{x},\partial_{\mathbf{x}})u(\mathbf{x})=rac{1}{(2\pi)^d}\int_{\mathbb{R}^d}e^{\imath\mathbf{x}\cdot\boldsymbol{\xi}}p(\mathbf{x},\boldsymbol{\xi})\hat{u}(\boldsymbol{\xi})d\boldsymbol{\xi},\ \boldsymbol{\xi}\in\mathbb{R}^d$$

p is the symbol of the operator \mathcal{P} , and is a smooth function of (x, ξ)

Symbol of the operator \mathcal{P}

$$p(\mathbf{x}, \boldsymbol{\xi}) = \sum_{|\boldsymbol{\alpha}| \leq m} a_{\boldsymbol{\alpha}}(\mathbf{x}) \boldsymbol{\xi}^{\boldsymbol{\alpha}}, \quad \boldsymbol{\xi}^{\boldsymbol{\alpha}} = \xi_1^{\alpha_1} \dots \xi_d^{\alpha_d}$$

The highest order term is the principal symbol. We use classical symbols S^m_{cl}

$$\left|\partial^{oldsymbol{eta}}_{oldsymbol{x}} \pmb{\mathit{p}}(oldsymbol{x},oldsymbol{\xi})
ight| \leq C(oldsymbol{lpha},oldsymbol{eta})(1+|oldsymbol{\xi}|)^{m-|oldsymbol{lpha}|}, \quad orall (oldsymbol{x},oldsymbol{\xi})\in \mathcal{K} imes \mathbb{R}^d,$$

Notations:

$$\mathcal{P} = \mathsf{Op}(p) \in \mathsf{OPS}^m, \quad p \in \mathcal{S}^m_{\mathsf{cl}} \Leftrightarrow \mathcal{P} \in \mathsf{OPS}^m, \quad \mathsf{OPS}^{-\infty} = \bigcap_{m \in \mathbb{R}} \mathsf{OPS}^m$$

PhD defense

Derivation of the DtN symbol - Helmholtz case

Nirenberg's factorization theorem: there exists $(\Lambda^+,\Lambda^-)\in \mathsf{OPS}^1$

$$\mathcal{L} = (\partial_x + i\Lambda^-) (\partial_x + i\Lambda^+) \mod \mathsf{OPS}^{-\infty}$$

= $\partial_x^2 + i (\Lambda^+ + \Lambda^-) \partial_x + i \mathrm{Op} \{\partial_x \lambda^+\} - \Lambda^- \Lambda^+ \mod \mathsf{OPS}^{-\infty}.$

Identify with the Helmholtz operator and get a system for (Λ^+, Λ^-)

$$\begin{cases} \Lambda^{+} + \Lambda^{-} = -i\rho_{0}^{-1}\partial_{x}(\rho_{0}) \\ \left(\Lambda^{+}\right)^{2} + i\rho_{0}^{-1}\partial_{x}(\rho_{0})\Lambda^{+} + i\operatorname{Op}\left\{\partial_{x}\lambda^{+}\right\} = \omega^{2}c_{0}^{-2} + \rho_{0}^{-1}\partial_{y}\left(\rho_{0}\partial_{y}\right) \end{cases}$$

 \rightarrow "One-way" reformulation of the equation "High frequency" asymptotic expansion for the total symbol λ^+

$$\lambda^+ \sim \sum_{j=-1}^{\infty} \lambda^+_{-j} = \lambda^+_1 + \lambda^+_0 + \lambda^+_{-1} + \cdots$$
 (classical symbol expansion)

Each symbol λ_{-j}^+ is homogeneous in $(\omega, \xi)^{-j}$. Once λ_1^+ is fixed, the expansion is unique and can computed formally

$$\lambda_{1}^{+} = \sqrt{\omega^{2}c_{0}^{-2} - \xi^{2}}, \ \lambda_{0}^{+} = -\imath \left(rac{\partial_{x}\left(
ho_{0}
ight)}{2
ho_{0}} + rac{\xi\partial_{y}\left(
ho_{0}
ight)}{2
ho_{0}\lambda_{1}^{+}} + rac{\omega^{2}\partial_{x}\left(c_{0}^{-2}
ight)}{4\left(\lambda_{1}^{+}
ight)^{2}} + rac{\xi\omega^{2}\partial_{y}\left(c_{0}^{-2}
ight)}{4\left(\lambda_{1}^{+}
ight)^{3}}
ight)$$

Microlocal zones

Hyperbolic zone: $\Re(\lambda_1^+) > 0 \rightarrow$ outgoing propagative waves Elliptic zone: $\Im(\lambda_1^+) < 0 \rightarrow$ outgoing evanescent waves

Rotation of the square-root branch-cut [Milinazzo et al. 1997]

$$\lambda_1^+ = e^{\imath lpha/2} \sqrt{e^{-\imath lpha} (\omega^2 c_0^{-2} - \xi^2)}, \ \alpha \in [0, -\pi], \ (+\imath \omega t \ \text{convention})$$

The branch-cut is rotated by an angle α in the complex plane

- Hyperbolic zone: $\omega c_0^{-1} > |\xi| \rightarrow \alpha = 0$,
- Elliptic zone: $\omega c_0^{-1} < |\xi| \rightarrow \alpha = -\pi$,
- Grazing zone: $\omega c_0^{-1} = |\xi|$, ill-posed problem

In practice we choose $\alpha \in (0, -\pi/2]$

 \rightarrow trade-off to capture both propagative and evanescent waves

Summary - building the DtN approximation

Approximate DtN operator with the M first symbols

$$\partial_{\mathbf{n}} u = -i \Lambda_{M}^{+} u, \quad \Lambda_{M}^{+} = \sum_{j=-1}^{M-2} \operatorname{Op}\left(\lambda_{-j}^{+}\right)$$

Take the trace on the boundary Σ to get $\widetilde{\Lambda}^+_{M},$ such that

$$\left(\widetilde{\Lambda}^+ - \widetilde{\Lambda}^+_M\right) \in \mathsf{OPS}^{1-M}$$

Technical difficulties

- the formal computation of λ⁺_{-i} can be involved (PDE dependent)
- the operator $\mathrm{Op}\left(\lambda^+_{-j}
 ight)$ is in general not unique and still non-local
- limited a priori to smooth variations of ho_0 and c_0

Next step - original contribution

Apply the theory to flow acoustics

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DtN symbol computation

Step 1 Nirenberg's factorization theorem for the flow acoustics operator

$$\mathcal{L} = rac{D_0}{Dt} \left(rac{1}{c_0^2} rac{D_0}{Dt}
ight) -
ho_0^{-1}
abla \cdot (
ho_0
abla), \quad rac{D_0}{Dt} = \imath \omega + \mathbf{v}_0 \cdot
abla$$

We note $M_x = v_{0,x}/c_0$, $M_y = v_{0,y}/c_0$, $M = ||\mathbf{v}_0||/c_0$, $k_0 = \omega/c_0$ **Step 2** Derive an operator equation for the outgoing characteristic Λ^+

$$\left(1-M_{x}^{2}\right)(\Lambda^{+})^{2}-\imath\left(\mathcal{A}_{1}+\mathcal{A}_{0}\right)\Lambda^{+}+\imath(1-M_{x}^{2})\mathsf{Op}\left\{\partial_{x}\lambda^{+}\right\}=\mathcal{B}_{2}+\mathcal{B}_{1}\;\mathsf{mod}\;\;\mathsf{OPS}^{-\infty}$$

with $\imath (\Lambda^+ + \Lambda^-) = (\mathcal{A}_1 + \mathcal{A}_0) / (M_x^2 - 1)$, $\sigma(\mathcal{A}_j), \sigma(\mathcal{B}_j) \in \mathcal{S}^j_{cl}, j = \{0, 1, 2\}$ **Step 3** Classical symbol expansion, identify 2nd order terms in (ω, ξ)

DtN principal symbol for flow acoustics

$$\lambda_{1}^{\pm} = \frac{1}{1 - M_{x}^{2}} \left[-M_{x} \left(k_{0} - \xi M_{y} \right) \pm \sqrt{k_{0}^{2} - 2k_{0}M_{y}\xi - (1 - M^{2})\xi^{2}} \right]$$

Microlocal zones delimited by $\omega c_0^{-1} = (M_y \pm \sqrt{1 - M_x^2}) \xi$ Subsonic flow M < 1 \Rightarrow two characteristic lines of opposite sign

DtN symbol computation

Step 3 - **bis** Compute the next symbols of the expansion ...

$$\lambda_0^+ = rac{\sigma(\mathcal{B}_1) + i\sigma(\mathcal{A}_0)\lambda_1^+ + i(1-M_x{}^2)\left(\partial_\xi\lambda_1^+\partial_y\lambda_1^+ - \partial_x\lambda_1^+
ight)}{2\sqrt{k_0^2 - 2k_0M_y\xi - (1-\mathsf{M}^2)\,\xi^2}}.$$

If $M_y = 0$ we have the simplification

Zeroth order symbol for an x-oriented flow

$$\lambda_0^+ = -\imath \frac{\partial_x(\rho_0)}{2\rho_0} \frac{k_0^2 - \xi^2}{k_0^2 - (1 - M_x^2)\xi^2} + \imath \frac{\partial_x(c_0)}{2c_0} \frac{k_0^2 + M_x^2 \xi^2}{k_0^2 - (1 - M_x^2)\xi^2}$$

$$\lambda_{-1}^+=...,\quad \lambda_{-2}^+=...$$

Going back to the operator level

Final step go back to the operator level

Approximate DtN operator

$$\begin{split} \widetilde{\Lambda}_{1}^{+} &= \mathsf{Op}(\lambda_{1}^{+}) = \frac{k_{0}}{1 - M_{n}^{2}} \left(-M_{n} + \imath M_{n} M_{\tau} \frac{\nabla_{\Sigma}}{k_{0}} + \sqrt{1 + X} \right) \\ X &= -2\imath M_{\tau} \frac{\nabla_{\Sigma}}{k_{0}} + \left(1 - \mathsf{M}^{2} \right) \frac{\Delta_{\Sigma}}{k_{0}^{2}}, \quad \mathsf{M} = \left\| \mathbf{v}_{0} \right\| / c_{0} \end{split}$$

For the half-space problem with constant coefficients

$$\widetilde{\Lambda}^+ = \widetilde{\Lambda}^+_1 \mod {\mathsf{OPS}}^{-\infty}$$

For variable coefficients and/or in the tangent plane approximation (n, au)

$$\widetilde{\Lambda}^+ = \widetilde{\Lambda}^+_1 \mod {\mathsf{OPS}}^0$$

The choice for $\mathsf{Op}(\lambda_1^+)$ is not unique, but has λ_1^+ as principal symbol

Issue: the approximate DtN map $\widetilde{\Lambda}_1^+$ is still non-local \rightarrow we need a local representation for $\sqrt{1+X}$

Localization procedure

High-frequency approximation for $\sqrt{1+X}$, $X \to 0$ ($\omega \to +\infty$) Use complexified Padé or Taylor approximations (N, α) for

$$\Lambda(Z) = e^{i\alpha/2}\sqrt{1+Z}, \quad Z = [e^{-i\alpha}(1+X)-1],$$

with Z a surfacic second order differential operator on the boundary $\boldsymbol{\Sigma}$

Taylor approximation

$$\Lambda(Z) pprox e^{i\alpha/2} \sum_{\ell=0}^{N} {1/2 \choose \ell} \left(e^{-i\alpha} (1+X) - 1 \right)^{\ell}$$

Padé approximation

$$\Lambda(Z) \approx K_0(\alpha) + \sum_{\ell=1}^N A_\ell(\alpha) X (1 + B_\ell(\alpha) X)^{-1}$$

Resulting local Absorbing Boundary Conditions

Complex Padé approximants: $\widetilde{\Lambda}_1^+ \to ABC_1^{N,\alpha}$ Complex Taylor approximants: $\widetilde{\Lambda}_1^+ \to ABC_1^{T0,\alpha}$ and $ABC_1^{T2,\alpha}$

High-order FEM implementation

Discretization on a conformal, high-order H^1 -basis

Weak formulation for the linearized potential equation

$$\begin{aligned} \forall \mathbf{v} \in \mathbf{V} \subseteq H^1(\Omega), \\ \int_{\Omega} \left[\rho_0 \nabla u \cdot \overline{\nabla \mathbf{v}} - \frac{\rho_0}{c_0^2} \frac{D_0 u}{D t} \frac{\overline{D_0 \mathbf{v}}}{D t} \right] d\Omega + i \int_{\Sigma} \mathcal{G} u \overline{\mathbf{v}} \ d\Sigma = \int_{\Omega} f \overline{\mathbf{v}} d\Omega \end{aligned}$$

The boundary operator ${\mathcal G}$ takes the same form as in the Helmholtz case

$$\int_{\Sigma} \mathcal{G} u \,\overline{v} \, d\Sigma = \int_{\Sigma} e^{i\alpha/2} \rho_0 k_0 \sqrt{1+Z} \, u \,\overline{v} \, d\Sigma$$

with $Z = e^{-i\alpha} \left(1 - 2iM_\tau \frac{\nabla_{\Sigma}}{k_0} + (1-\mathsf{M}^2) \, \frac{\Delta_{\Sigma}}{k_0^2} \right) - 1$

High-order FEM implementation

Discretization on a conformal, high-order H^1 -basis

Weak formulation for the linearized potential equation

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with $Z = e^{-i\alpha} \left(1 - 2iM_{\tau} \frac{\nabla_{\Sigma}}{k_0} + (1 - M^2) \frac{\Delta_{\Sigma}}{k_0^2} \right) - 1$ Taylor approximants: $ABC_1^{T2,\alpha}$

$$\begin{split} \int_{\Sigma} \mathcal{G} u \, \overline{v} \, d\Sigma &= \cos(\alpha/2) \int_{\Sigma} \rho_0 k_0 \, u \, \overline{v} \, d\Sigma \\ &+ e^{-i\alpha/2} \left(\int_{\Sigma} \rho_0 M_\tau \nabla_{\Sigma} u \, \overline{v} \, d\Sigma - \int_{\Sigma} \rho_0 \frac{(1-\mathsf{M}^2)}{2k_0} \nabla_{\Sigma} u \, \nabla_{\Sigma} \overline{v} \, d\Sigma \right) \end{split}$$

High-order FEM implementation - Padé case

Padé approximants: $ABC_1^{N,\alpha}$

$$\int_{\Sigma} \mathcal{G} u \, \overline{v} \, d\Sigma = \int_{\Sigma} \rho_0 k_0 \mathcal{K}_0(\alpha) \, u \, \overline{v} \, d\Sigma + \sum_{\ell=1}^N \int_{\Sigma} \rho_0 k_0 \mathcal{A}_\ell(\alpha) \, X \, \varphi_\ell \, \overline{v} \, d\Sigma$$

with $X = -2\imath M_{\tau} rac{
abla_{\Sigma}}{k_0} + \left(1 - \mathsf{M}^2\right) rac{\Delta_{\Sigma}}{k_0^2}$

Introduce auxiliary fields $\varphi_{\ell} = (1 + B_{\ell}X)^{-1}u$ to obtain a sparse discretization of inverse operators

 \rightarrow augmented system of *N* surfacic equations

$$\forall \ell \in [1, N], \ \forall v_{\ell} \in H^1(\Sigma)^N, \quad \int_{\Sigma} (1 + B_{\ell}(\alpha) X) \varphi_{\ell} \overline{v}_{\ell} \ d\Sigma = \int_{\Sigma} u \overline{v}_{\ell} \ d\Sigma$$

Global linear system of size $[N_{dof,\Omega} + (N \times N_{dof,\Sigma})]$

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- 5. The 3D turbofan intake radiation problem Numerical results for the turbofan problem Solver weak scalability

Longitudinal heterogeneous waveguide

Example 1: No mean flow, single mode propagation in a heterogeneous waveguide: $\rho_0 = 1$, $c_0(x, y) = c_0(x)$



Single mode analytic solution for a linear profile

$$u_{\mathrm{ex}}^{n}(x,y) = \cos\left(k_{y}y\right) \operatorname{Ai}\left(e^{-\frac{2i\pi}{3}}\frac{k_{y}^{2}-\omega^{2}(ax+b)}{(a\omega^{2})^{2/3}}\right), \quad k_{y} = \frac{n\pi}{H}, \quad n \in \mathbb{N}$$

Approximation at the symbol level

 $c_0^{-2}(x) = 5x + 0.1$, L = 1 at fixed frequency $\omega = 30$

Analytic total symbol $\lambda^{+} = -\imath e^{-\frac{2\imath\pi}{3}} \left(a\omega^{2}\right)^{1/3} \frac{\operatorname{Ai'}(\zeta)}{\operatorname{Ai}(\zeta)}, \quad \zeta = e^{-\frac{2\imath\pi}{3}} \frac{\xi^{2} - \omega^{2}(aL+b)}{(a\omega^{2})^{2/3}}$



• singularity in the grazing regime $\xi \approx \omega c_0^{-1}$

Numerical results



Use the derivative of $c_0 \Rightarrow$ more accurate ABC: $ABC_2^{N+1,\alpha} > ABC_1^{N,\alpha}$

Gain of \approx two order of magnitude Precision limited by the DtN symbol truncation

Similar conclusion with Taylor approximations of order 0 and 2

Point source convection in free field

Example 2: $\rho_0 = c_0 = 1$, point source in uniform mean flow of angle θ Padé approximants $\rightarrow ABC_1^{N,\alpha}$, Taylor approximants $\rightarrow ABC_1^{T0,\alpha}$, $ABC_1^{T2,\alpha}$ Attempt to incorporate curvature effects from λ_0^+ (circle of radius *R*)

$$ABC_2 = ABC_1 + (1 - M^2)/(2R)$$



 $\mathsf{ABC}_1^{\mathsf{N},\alpha}$ is robust for high Mach numbers - wavelength ratio $(1+\mathsf{M})/(1-\mathsf{M})$

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Waveguide problem with straight partitions

Domain partition:
$$\Omega = \bigcup_{i=0}^{N_{dom}-1} \Omega_i, \ \Sigma_{ij} = \overline{\partial \Omega_i \cap \partial \Omega_j}, \ j \neq i$$

$$\begin{array}{c} & & \\ & &$$

Two test cases

- 1. Linear speed of sound profile : $c_0^{-2}(x) = ax + b$
- 2. Transverse density and/or speed of sound : $c_0(y), \rho_0(y)$

Longitudinal heterogeneous waveguide

Reminder - ABC study

- $ABC_2^{N+1,\alpha} > ABC_1^{N,\alpha}$
- Precision limited by the truncation of the total DtN symbol λ^+

Plot theoretical convergence radius $ho(m,x) = \sqrt{\left|\mathcal{T}_{ij}^m\mathcal{T}_{ji}^m\right|}$ at $\omega = 30$



DDM waveguide study - input boundary condition with the 21 first modes

N _{dom}	$ABC_1^{T0,\alpha}$	$ABC_1^{T2,\alpha}$	$ABC_1^{6,\alpha}$	$ABC_2^{6,\alpha}$
8	76 (dnc)	51 (87)	34 (37)	24 (27)

Table: GMRES(Jacobi) iterations to 10^{-6} at $\omega = 40, \alpha = -\pi/4, d_{\lambda} = 12$

Transverse heterogeneous waveguide

Gaussian waveguide:
$$c_0(y) = 1.25 \left(1 - 0.4 e^{-32(y-H/2)^2}
ight),
ho_0(y) = c_0^2(y)$$



Figure: Theoretical eigenvalues of the DDM iteration matrix, $\omega = 50$

- Usual Padé approximation (classical principal symbol) $S_i = ABC_{k_0}^{N,\alpha}$
- New approximation (semi-classical principal symbol) S_i = ABC^{N,α}_{ω,S} → almost perfect clustering

Illustration for a Gaussian waveguide

DDM - large PML on Γ_{∞} - input mode n = 4 - $N_{dom} = 8$



Table: GMRES iterations to 10^{-6} at $\omega = 160, d_{\lambda} = 24$

Convergence in N_{dom} iterations \rightarrow continuous block LU factorization

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Convected problem in freefield - circular interfaces



Numerical solution after 4 GMRES iterations. Parameters: M = 0.9, $\theta = \pi/4$, p = 9, $d_{\lambda} = 8$, $N_{dom} = 5$, $\omega = 6\pi$. L^2 -error "PML-analytical solution" : 0.8%.

Padé conditions are robust for high Mach numbers

Mach number variation - circular interfaces



Figure: GMRES iterations to reach 10^{-6} for $N_{\rm dom}=5,~\omega=6\pi$

- Only Padé conditions are robust for high Mach numbers
- Layered partition: iteration number increases as O(N_{dom})

Convected problem in freefield - arbitrary decomposition

Automatic partitioning

- Cross-points \rightarrow harder to design ABCs
- Industrial need good load balancing between subproblems
- Shorter connectivity graph $\mathcal{O}(\sqrt{N_{\text{dom}}})$



Corner problem : use Padé approximants on edges with Sommerfeld-type condition on corners \rightarrow approximate corner treatment Resulting condition $\mathsf{ABC}^{N,\alpha}_S$

Are such Padé conditions still effective for domain decomposition ?

Numerical study - arbitrary decomposition



GMRES iterations to reach 10^{-6} for $\omega = 6\pi, d_{\lambda} = 8$

- lost of robustness for high Mach numbers
- ABC^{T2,- $\pi/2$}, ABC^{4,- $\pi/2$} \rightarrow similar performance
- ABC^{T2,-π/2} is cheaper (and easier) to implement



$$M = 0.7, N_{dom} = 16, ABC^{T2, -\pi/2}$$

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The boundary value problem

Given a flight configuration (mean flow), predict the radiated noise from the fan, at multiples of the blade passing frequency $\omega_{bpf}/(2\pi) = 1300 \text{ Hz}$



The mean flow is pre-computed and interpolated on the acoustic mesh

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DDM for the 3D turbofan problem





Parallel GmshDDM solver (mono-thread)

From $N_{\rm dom} > 128$, under 10 minutes and less than 3Gb per process lterations for $N_{\rm dom} = 64$

- ABC^{T2, $-\pi/2$}: 372 GMRES iterations to 10^{-6}
- ABC^{T0,0}: > 2000 GMRES iterations to reach 10^{-3}

DDM for the 3D turbofan problem

 $2 \times \omega_{bpf}$ - $N_{dofs} = 73M$ - nnz = 5B \approx 50 wavelengths in Ω Parallel GmshDDM solver (mono-thread), $N_{dom} = \#$ MPI = 128 \rightarrow 2hours 3min, 26 Gb peak memory, 712 GMRES iterations (with lining)



Figure: Real part of the acoustic velocity potential for the mode (48,1) at $2 \times \omega_{bpf}$ (2600 Hz) without (left) and with (right) acoustic lining treatment.

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Weak scalability assessment

3D Helmholtz problem with $d_{\lambda} = 7.5$ points per wavelength



Figure: Weak scaling timing for 1 iteration

Limitations

- memory load balancing: [20-34] Gb for 1024 processes
- number of iterations scales as $\mathcal{O}(N_{\text{dom}}^{1/3})$ in 3D

PhD defense

Distributed memory solver for high frequency flow acoustics

- 1. Extension of a generic domain decomposition method to flow acoustics
- 2. Development of non-reflecting boundary conditions: ABCs and PMLs
 - extension to media with heterogeneities and convection
 - general PML procedure for flow acoustics
- 3. Code development (GmshDDM), validation and assessment of ABCs in a domain decomposition context
- 4. Proof of concept turbofan intake
- 5. 80% scalability up to 700M high-order unknowns and 1024 MPI processes

Limitations

- The iterative solver does not scale with $N_{\rm dom} \rightarrow$ requires coarse space
- Theoretical limitations corners, curved boundaries, PML-DDM, etc.

Future developments

- Extension to Pierce equation \rightarrow turbofan exhaust [Spieser, Bailly 2020]
- Modern discretization techniques such as HDG or HHO [Li et al. 2013]
- Interfaces through the mesh \rightarrow immersed transmission conditions

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Thank you !